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(54) **HYBRID BEARING TURBOMACHINE**

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CPC **F04D 29/051** (2013.01); **F04D 29/058**
(2013.01); **F04D 29/059** (2013.01)

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CPC F01D 25/162; F01D 25/164
See application file for complete search history.

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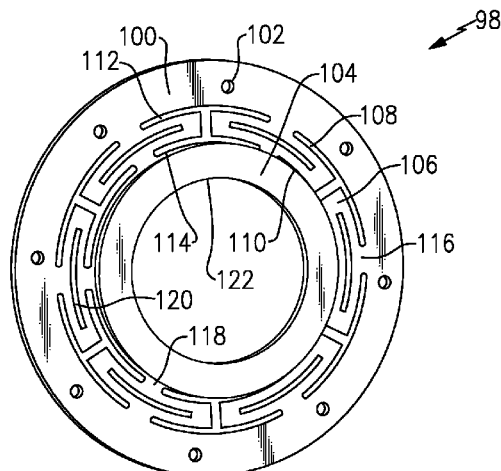
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ABSTRACT

A turbomachine includes a housing and a rotatable shaft, where at least a portion of the rotatable shaft is located in the housing. The turbomachine also includes a magnetic thrust bearing that axially positions the rotatable shaft and a radial bearing that centers the rotatable shaft. The turbomachine includes a flexure including a first portion secured to the housing and a second portion axially moveable relative to the first portion. The second portion of the flexure is connected to the radial bearing, and the second portion moves axially to eliminate thrust loads on the radial bearing and allow the magnetic thrust bearing to carry axial loads.

21 Claims, 3 Drawing Sheets



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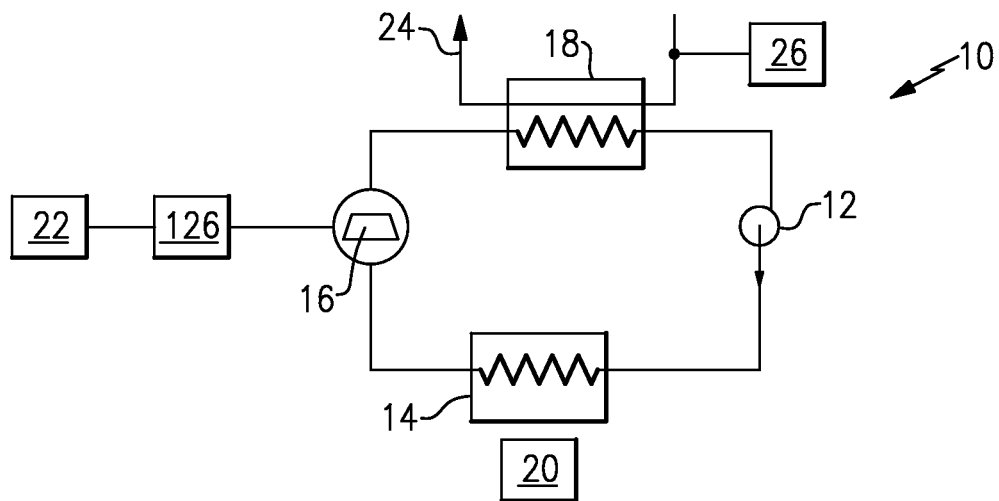


FIG. 1

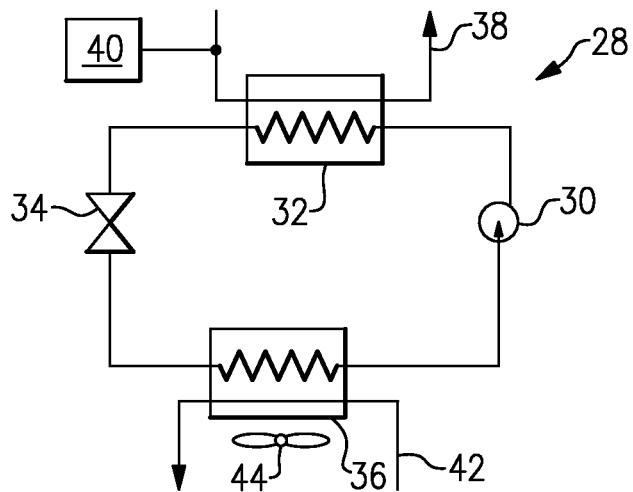
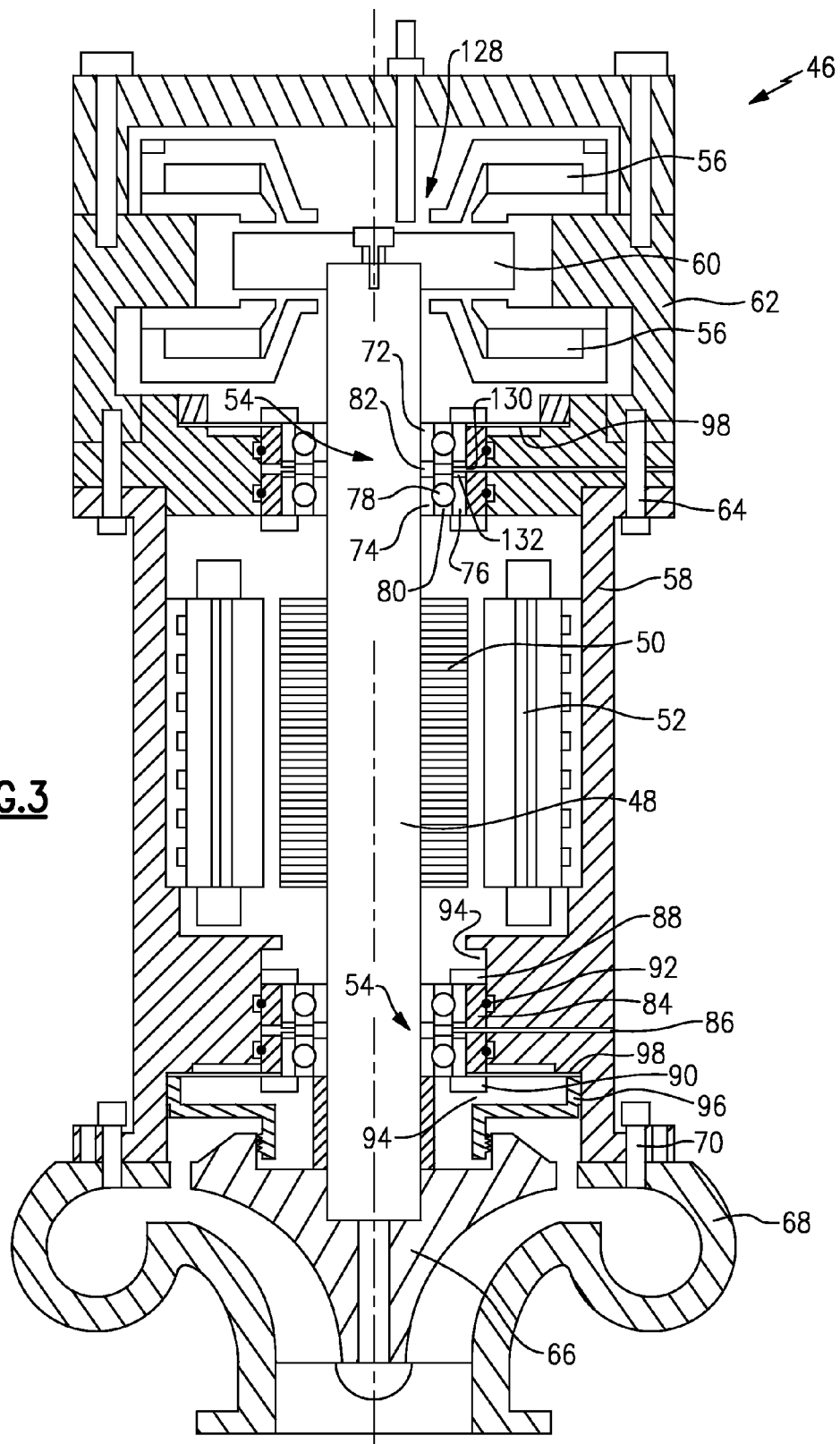


FIG. 2

FIG. 3



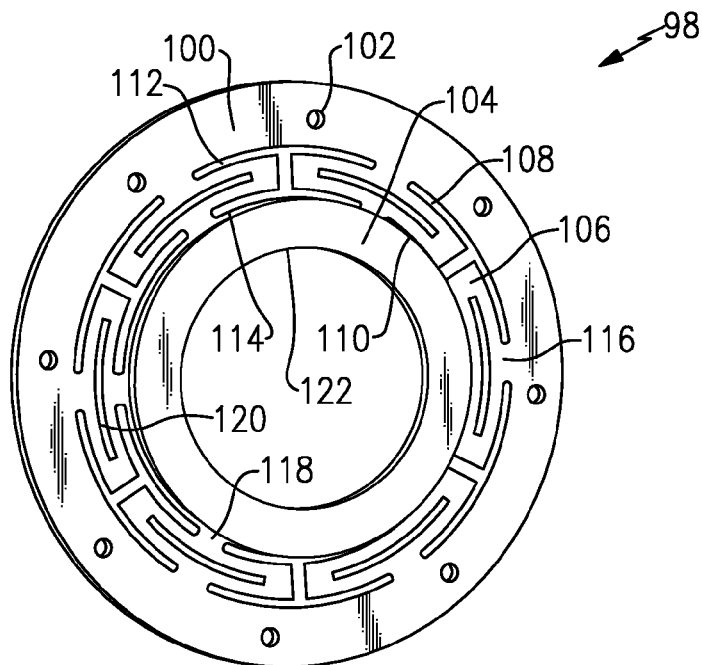


FIG. 4

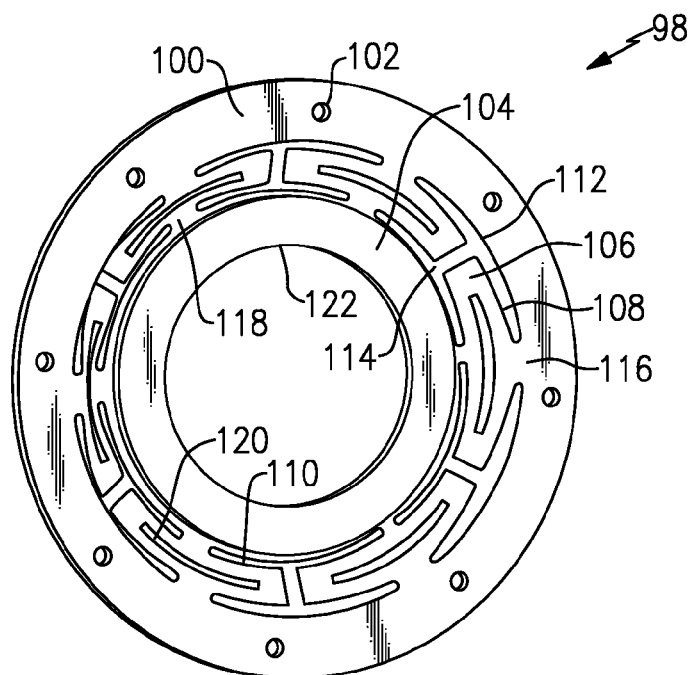


FIG. 5

1

HYBRID BEARING TURBOMACHINE**BACKGROUND OF THE INVENTION**

High speed turbomachines can employ various types of bearings. For example, rolling element bearings, magnetic bearings, foil bearings or fluid film bearings can be employed.

Oil free turbomachines provide many benefits. For one, there are more effective heat transfer surfaces and no issues with oil stability or contamination. Oil free turbomachines have employed magnetic bearings, hydrostatic bearings, or foil bearings. Turbomachines have also been developed using refrigerant lubricated roller bearings and hydrostatic bearings.

Disadvantages of current turbomachines are the expense and complexity of the magnetic bearing, as well as scaling problems with foil bearings that makes large machines unfeasible. Additionally, the power consumption of hydrostatic bearings can be high, and when operated under high loads, the reliability of ball bearings can decrease.

SUMMARY OF THE INVENTION

A turbomachine includes a housing and a rotatable shaft, where at least a portion of the rotatable shaft is located in the housing. The turbomachine also includes a magnetic thrust bearing that axially positions the rotatable shaft and a radial bearing that centers the rotatable shaft. The turbomachine includes a flexure including a first portion secured to the housing and a second portion axially moveable relative to the first portion. The second portion of the flexure is connected to the radial bearing, and the second portion moves axially to eliminate thrust loads on the radial bearing and allow the magnetic thrust bearing to carry axial loads.

Another turbomachine includes a housing and a rotatable shaft. At least a portion of the rotatable shaft is located in the housing, and the rotatable shaft is substantially vertical when the turbomachine is in use. The turbomachine includes a magnetic thrust bearing that axially positions the rotatable shaft and a radial bearing that centers the rotatable shaft. The turbomachine is employed in a system that employs a working fluid, and the working fluid lubricates the radial bearing. The turbomachine includes a flexure including a first portion secured to the housing and a second portion axially moveable relative to the first portion. The second portion of the flexure is connected to the radial bearing, and the second portion moves axially to reduce thrust loads on the radial bearing and allow the magnetic thrust bearing to carry axial loads.

A method of reducing thrust loads on a radial bearing of a turbomachine includes the steps of rotating a shaft located in a housing of the turbomachine and axially positioning the shaft with a magnetic thrust bearing. The method further includes the steps of securing a first portion of a flexure to the housing and axially moving a second portion of the flexure relative to the first portion of the flexure to axially translate the radial bearing to eliminate axial loads on the radial bearing and allow by the magnetic thrust bearing to carry axial loads.

These and other features of the present invention will be best understood from the following specification and drawings.

BRIEF DESCRIPTION OF THE DRAWINGS

The various features and advantages of the invention will become apparent to those skilled in the art from the following

2

detailed description of the currently preferred embodiment. The drawings that accompany the detailed description can be briefly described as follows:

FIG. 1 schematically illustrates an organic rankine cycle;

FIG. 2 schematically illustrates a refrigeration system;

FIG. 3 schematically illustrates a cross-sectional view of a turbomachine;

FIG. 4 schematically illustrates a flexure of the turbomachine in an initial position; and

FIG. 5 schematically illustrates the flexure of the turbomachine in an extended position.

DETAILED DESCRIPTION OF THE PREFERRED EMBODIMENTS

FIG. 1 illustrates an example organic rankine cycle 10 that converts heat into work. The organic rankine cycle 10 includes a pump 12, an evaporator 14, a turbine 16, and a condenser 18. An organic fluid circulates through the closed circuit system. The fluid flows as a liquid through the pump 12 and is pumped from a low pressure to a high pressure. The high pressure liquid then flows through the evaporator 14, where the fluid is heated and boiled at a constant pressure by an external heat source 20, and the fluid becomes a vapor. The vapor flows through the turbine 16 and is expanded, generating power. The shaft power generated by the turbine 16 is fed to a generator 126, which converts the shaft power to electrical power which is provided to an electric power grid 22 or load. The temperature and the pressure of the vapor decreases. The vapor then enters the condenser 18, where the fluid rejects heat to a cooling medium 24, heating the cooling medium 24. A pump or fan 26 moves the fluid medium 24 through the condenser 18. The fluid is condensed to become a saturated liquid. The liquid then flows through the pump 12, completing the cycle.

FIG. 2 illustrates an example vapor compression system 28 that includes a compressor 30, a condenser 32, an expansion device 34, and an evaporator 36. Refrigerant circulates through the closed circuit system. The refrigerant exits the compressor 30 at a high pressure and a high enthalpy. The refrigerant then flows through the condenser 32 at a high pressure. A fluid medium 38, such as water or air, flows through the condenser 32 and accepts heat from the refrigerant flowing through the condenser 32. The refrigerant exits the condenser 32 at a low enthalpy and a high pressure. A water pump or fan 40 moves the fluid medium 38 through the condenser 36. The refrigerant then passes through the expansion device 34, which regulates the flow of the refrigerant. The refrigerant then flows through the evaporator 36 and exits at a high enthalpy and a low pressure. In the evaporator 36, the refrigerant accepts heat from a fluid medium 42, heating the refrigerant. In one example, the fluid medium 42 is air, and a fan 44 blows the fluid medium 42 through the evaporator 36. The refrigerant then reenters the compressor 30, completing the cycle.

FIG. 3 schematically illustrates a cross-sectional view of a turbomachine 46. The above-described turbine 16 and compressor 30 are both turbomachines 46. The turbomachine 46 can also be a gas expander or a steam turbine. The below-mentioned and described components of the turbomachine 46 are circumferential in shape.

The turbomachine 46 includes a rotatable shaft 48, a rotor 50, a stator 52, and two sets of radial bearings 54 housed in a main housing 58. Each set of radial bearings 54 can be mounted in bearing units. The stator 52 is fixed inside the main housing 58, and the rotor 50 is fixed to and rotates with the rotatable shaft 48. The sets of radial bearings 54 allow for

smooth rotation of the rotor **50** and are lubricated with a liquid working fluid of the turbomachine **46**. In one example, the working fluid is refrigerant.

If the turbomachine **46** is a compressor **30**, rotation of the rotatable shaft **48** compresses the refrigerant to a high pressure. If the turbomachine **46** is a turbine **16**, rotation of the rotatable shaft **48** expands the vapor, providing power.

A magnetic thrust bearing **128** includes a magnetic thrust bearing housing **62** contains a rotating thrust disc **60** located between magnetic actuators **56**. The rotating thrust disc **60** is attached to an end of the rotatable shaft **48** and is located between the magnetic actuators **56**. The magnetic thrust bearings **128** manage axial, or thrust, loads. The magnetic thrust bearing **128** locates the rotatable shaft **48** axially by providing forces countering the force from an impeller **66** and gravity, as well as other forces. The main housing **58** and magnetic thrust bearing housing **62** are attached with a fastener **64**. Magnetic thrust bearings **128** are less complicated than radial bearings **54**, and their load capacity is better than other types of oil free bearings.

An impeller **66** is attached to an opposing end of the rotatable shaft **48** and is located in an impeller housing **68**. The main housing **58** and the impeller housing **68** are attached with a fastener **70**.

Each set of radial bearings **54** includes two ball bearings **72**. Each ball bearing **72** includes a circular inner race **74**, a circular outer race **76**, and a plurality of balls **78** encased in a cage **80** located between the circular inner race **74** and the circular outer race **76**. Each ball bearing **72** is in contact with the rotatable shaft **48**. In one example, the plurality of balls **78** are ceramic.

A spacer **82** that contacts the rotatable shaft **48** is located between the circular inner races **74** of the two ball bearings **72** of each set of radial bearings **54**, and a spacer **130** is located between the circular outer races **76** of the two ball bearings **72**. The spacer **130** includes a passage **132** that allows the working fluid to flow to the ball bearings **72**. A bearing housing **84** including a portion of a working fluid passage **86** is located radially outwardly from each set of ball bearings **72**. The working fluid passage **86** allows the working fluid to flow to cool and lubricate the ball bearings **72**.

A first travel stop **88** is located on one side of the bearing housing **84**, and a second travel stop **90** is located on an opposing side of the bearing housing **84**. The travel stops **88** and **90** limit the axial translation/travel of the ball bearings **72**, and a gap **94** is defined axially outwardly from the travel stops **88** and **90**. The main housing **58** is located radially outwardly from the bearing housing **84** and also defines a portion of the working fluid passage **86** that provides the working fluid to the radial bearings **54**. A seal **92**, for example an O-ring, provides a seal between the bearing housing **84** and the main housing **58** to prevent leakage of the working fluid that flows through the working fluid passage **86** between the main housing **58** and the bearing housing **84**.

FIGS. **4** and **5** illustrate a flexure **98** of the turbomachine **46**. FIG. **4** shows the flexure **98** when flat in an initial position, and FIG. **5** shows the flexure **98** in an extended position when extended to eliminate or reduce thrust loads from the radial bearings **54** that are caused by the translation of the rotatable shaft **48**. One flexure **98** is associated with each set of radial bearings **54**. In one example, the flexure **98** is formed of laser cut sheet metal. In one example, the sheet metal is carbon steel and a stiffness of 100 pounds force per millimeter. The flexure **98** is an axially flexible mount in which a radial bearing **54** is mounted to eliminate thrust loads. The flexure **98** can also provide position feedback for the magnetic thrust bearing **128**.

The flexure **98** includes an outer ring **100** including a plurality of apertures **102**, an inner ring **104**, and a plurality of flexible attachment features **106** that connect the outer ring **100** to the inner ring **104**. The inner ring **104** is connected to the radial bearings **54**. In one example, the flexure **98** includes six attachment features **106**. Each attachment feature **106** is substantially quadrilateral in shape, and an outer edge **108** and an inner edge **110** of each attachment feature **106** have a curvature that matches the curvature of an inner edge **112** of the outer ring **100** and an outer edge **114** of the inner ring **104**, respectively. The outer edge **108** of the attachment feature **106** is attached to the outer ring **100** with an outer flange **116**, and the inner edge **110** of the attachment feature **106** is attached to the inner ring **104** with an inner flange **118**. Each attachment feature **106** includes an aperture or slot **120** within the quadrilateral shape to reduce axial stiffness.

Returning to FIG. **3**, the outer ring **100** of the flexure **98** is located between the main housing **58** and an end ring **96** and secured to the main housing **58** by fasteners (not shown) received in the apertures **102**. The inner ring **104** of the flexure **98** is located between the bearing housing **84** and the second travel stop **90**. An inner edge **122** of the inner ring **104** of the flexure **98** contacts an outer diameter of one of the ball bearings **72**. The attachment features **106** are located in the gap **94** between the main housing **58** and the end ring **96** that is sized to allow for movement of the attachment features **106**.

The magnetic thrust bearing **128** carries all the axial loads because the radial bearings **54** are allowed to float axially. When this occurs, the inner ring **104** of the flexure **98** moves relative to the secured outer ring **100**, eliminating thrust loads on the radial bearings **54**. In one example, the inner ring **104** and the outer ring **100** can displace and deform approximately 0.5 millimeters (0.02 inches) relative to each other.

In another example, the radial bearing **54** is a foil bearing or a journal bearing (not shown). In this example, the radial bearings **54** float freely in the axial direction, allowing the magnetic bearing **126** to carry the axial load.

In one example, the turbomachine **46** is positioned substantially vertical or vertical during use such that the rotatable shaft **48** is substantially vertical or vertical. When the rotatable shaft **48** is substantially vertical or vertical, there are very small or no radial loads on the radial bearings **54** during start up or operation of the turbomachine **46**.

In another example, the turbomachine **46** includes an actively controlled hydrostatic axial bearing. The radial bearings **54** mounted in the flexure **98** supports the turbomachine **46** during start up, and then the axial load is absorbed by the hydrostatic axial bearing.

The turbomachine **46** provides a cost and performance savings. By employing a hybrid bearing system with a magnetic thrust bearing **128** and radial bearings **54** mounted to float axially, a smaller turbomachine **46** with higher thrust loads can be produced. A hybrid bearing turbomachine **46** including both a magnetic thrust bearing **128** and radial bearings **54** can also have a tighter clearance than turbomachines having only magnetic bearings.

The foregoing description is only exemplary of the principles of the invention. Many modifications and variations of the present invention are possible in light of the above teachings. The preferred embodiments of this invention have been disclosed, however, so that one of ordinary skill in the art would recognize that certain modifications would come within the scope of this invention. It is, therefore, to be understood that within the scope of the appended claims, the invention may be practiced otherwise than as specifically

5

described. For that reason the following claims should be studied to determine the true scope and content of this invention.

What is claimed is:

1. A turbomachine comprising:

a housing;

a rotatable shaft, wherein at least a portion of the rotatable shaft is located in the housing;

a magnetic thrust bearing that axially positions the rotatable shaft;

a radial bearing that centers the rotatable shaft; and

a flexure including a first portion secured to the housing and a second portion axially moveable relative to the first portion, wherein the second portion contacts the radial bearing, and the second portion moves axially to eliminate thrust loads on the radial bearing and to allow the magnetic thrust bearing to carry axial loads, wherein the flexure is planar.

2. The turbomachine as recited in claim 1 wherein the radial bearing comprises at least one of a ball bearing, a foil bearing, and a journal bearing.

3. The turbomachine as recited in claim 1 including a bearing housing located radially outwardly of the radial bearing, and the bearing housing includes a portion of a working fluid passage to provide a working fluid to the radial bearing.

4. The turbomachine as recited in claim 3 wherein the housing includes another portion of the working fluid passage to provide the working fluid to the radial bearing.

5. The turbomachine as recited in claim 4 including a seal between the bearing housing and the housing.

6. The turbomachine as recited in claim 1 wherein the turbomachine is employed in a system that employs a working fluid, and the working fluid lubricates the radial bearing.

7. The turbomachine as recited in claim 1 wherein the flexure is formed of sheet metal.

8. The turbomachine as recited in claim 1 wherein the first portion of the flexure is an outer ring, the second portion of the flexure is an inner ring, and the flexure includes a plurality of flexible attachment features located therebetween that allow the inner ring to move relative to the outer ring, wherein each of the plurality of attachment features include an aperture to reduce axial stiffness.

9. The turbomachine as recited in claim 1 wherein the rotatable shaft is substantially vertical when the turbomachine is in use.

10. The turbomachine as recited in claim 1 wherein the turbomachine is a turbine in an organic rankine cycle.

11. The turbomachine as recited in claim 1 wherein the turbomachine is a compressor of a refrigeration system.

12. A turbomachine comprising:

a housing;

a rotatable shaft, wherein at least a portion of the rotatable shaft is located in the housing, and the rotatable shaft is substantially vertical when the turbomachine is in use;

a magnetic thrust bearing that axially positions the rotatable shaft;

6

a radial bearing that centers the rotatable shaft, wherein the turbomachine is employed in a system that employs a working fluid, and the working fluid lubricates the radial bearing; and

a flexure including a first portion secured to the housing and a second portion axially moveable relative to the first portion, wherein the second portion contacts the radial bearing, and the second portion moves axially to eliminate thrust loads on the radial bearing and to allow the magnetic thrust bearing to carry axial loads, wherein the flexure is planar.

13. The turbomachine as recited in claim 12 including a bearing housing located radially outwardly of the radial bearing, and the bearing housing includes a portion of a working fluid passage and the housing includes another portion of the working fluid passage, wherein the working fluid passage provides the working fluid to the radial bearing.

14. The turbomachine as recited in claim 12 wherein the first portion of the flexure is an outer ring, the second portion of the flexure is an inner ring, and the flexure includes a plurality of flexible attachment features located therebetween that allow the inner ring to move relative to the outer ring, wherein each of the plurality of attachment features include an aperture to reduce axial stiffness.

15. The turbomachine as recited in claim 12 wherein the turbomachine is a turbine in an organic rankine cycle.

16. The turbomachine as recited in claim 12 wherein the turbomachine is a compressor of a refrigeration system.

17. The turbomachine as recited in claim 12 wherein an axial portion of an impeller is controlled to minimize a gap between the impeller and an impeller housing to maximize a compression efficiency or an expansion efficiency.

18. A method of reducing thrust loads on a radial bearing of a turbomachine, the method comprising the steps of:

rotating a shaft located in a housing of a turbomachine;

axially positioning the shaft with a magnetic thrust bearing;

securing a first portion of a flexure to the housing, wherein the flexure is planar; and

axially moving a second portion of the flexure relative to the first portion of the flexure to axially translate radial bearing to eliminate axial loads on the radial bearing created by the magnetic thrust bearing and to allow the magnetic thrust bearing to carry axial loads.

19. The turbomachine as recited in claim 1 wherein the flexure has a constant thickness.

20. The turbomachine as recited in claim 1 wherein the flexure is made of carbon steel.

21. The turbomachine as recited in claim 1 wherein the flexure is laser cut.

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